Jet-Wake Thermal Characteristics of Heated Turbulent Jets in Crossflow

S. A. Sherif* University of Miami, Coral Gables, Florida 33124 R. H. Pletcher† Iowa State University, Ames, Iowa 50011

This paper is one in a series reporting on jets discharging to crossflowing water streams. The experiments were carried out in the 0.61-m × 1.067-m semiclosed circuit water channel at Iowa State University. Hot water was injected vertically upward from a circular pipe located near the channel bottom to simulate the turbulent heated jet. Contours of mean and root-mean-square temperatures both across and along the jet are reported up to lateral and downstream distances of 1.15 and 37 jet diam, respectively, and for velocity ratios of 1, 2, 4, and 7. The contours were used to explain the complex interaction mechanism between the freestream and the jet in both the jet and wake regions.

Nomenclature

D= jet diameter at discharge

M = jet-to-freestream velocity ratio, U_i/U_{∞}

= mean temperature

= jet discharge temperature

= freestream temperature

 T_j T_{∞} t'= fluctuating temperature

 U_j = jet discharge velocity

= freestream velocity

= streamwise distance

= transverse (spanwise) distance

= vertical distance

Introduction

N most practical situations, jets and plumes are either dis-Acharged vertically or at an angle to a crossflow. This is the case in exhaust gases discharging into the atmosphere, condenser cooling water pouring into rivers, and the jet in the V/STOL aircraft.

The words jet and plume are sometimes used interchangeably. A plume is, in general, a free shear flow in which the primary source of kinetic energy and momentum flux arises through body forces. In a jet the kinetic energy and momentum are large at discharge, having been imparted to the flow by a pressure drop through an orifice, nozzle, or tube. A forced plume or a buoyant jet are flows whose motion is in transition from a jet to a plume.1

Experimental work on the behavior of plumes in flowing ambients is numerous. Bryant and Cowdrey² studied the trajectory of heated air plumes in a wind tunnel. Slawson and Csanady³ conducted experiments to determine the mean path of buoyant bent-over chimney plumes. Halitsky4 visualized smoke plumes with a single camera. Hewett et al.5 and Hewett⁶ studied smokestack plumes in a stable atmosphere.

Bringfelt⁷ measured plume rise at industrial chimneys. The same problem was also studied by Leavitt et al., 8 Carpenter et al.,9 and Thomas et al. 10 Weil 11 and Barilla 12 injected ferric chloride into a water towing tank in order to model a plume in a laminar crossflow. Barilla varied the initial angle of the issuing plume and the bending angle of the discharge nozzle for different Reynolds numbers and observed that far downstream of the discharge nozzle the plume rise is independent of the orientation of the discharge.

Experiments on round turbulent nonbuoyant jets where the main thrust was on velocity and/or turbulence intensity measurements were conducted by Jordinson, ¹³ Shandorov, ¹⁴ Gordier, ¹⁵ Keffer, ¹⁶ Keffer and Baines, ¹⁷ Pratte, ¹⁸ Pratte and Baines, ¹⁹ Gerend, ²⁰ Patrick, ²¹ Platten and Keffer, ²² Margason, ²³ Bergeles et al., ²⁴ Storms, ²⁵ Gelb and Martin, ^{26,27} Crowe and Riesebieter, ²⁸ Chassaing et al., ²⁹ and Chan and Kennedy.30

Experiments on round turbulent buoyant jets in which the main thrust was on velocity, concentration, temperature, or trajectory measurements were carried out by Fan, 31,32 Abraham and Eysink,³³ Abraham,³⁴ Turner,³⁵ Vadot,³⁶ Ayoub,³⁷ Cavola, 38 and Cavola and Davis. 39

Experiments on round turbulent heated nonbuoyant jets in which the main thrust was on trajectory, velocity, temperature, or penetration measurements were carried out by Callaghan and Ruggeri, ^{40,41} Callaghan and Bowden, ⁴² Ruggeri and Callaghan, ⁴³ Ruggeri, ⁴⁴ Campbell and Schetz, ⁴⁵⁻⁴⁷ Payne and Schetz, ⁴⁸ Ramsey, ⁴⁹ Ramsey and Goldstein, ⁵⁰ Kamotani and Greber,⁵¹ and Wark and Foss.⁵²

Experiments on turbulent nonbuoyant jets in which the main thrust was on studying the structure of the jet wake, jet cross section, vortex shedding in the jet, or measuring the vorticity in the jet or wake were carried out by Chang-Lu and Hsiu-Chen,⁵³ Rouse,⁵⁴ McAllister,⁵⁵ Pratte and Keffer,⁵⁶ McMahon et al., 57 Fearn and Weston, 58 Weston, 59 Moussa, 60 and Moussa et al.61

Experiments on round turbulent nonbuoyant jets in which the main thrust was on pressure measurements were carried out by Vogler, 62 Peake, 63 Bradbury and Wood, 64 Wooler et al.,65 Hackett and Miller,66 McMahon and Mosher,67 Mosher,68 Wu et al.,69 Mikolowsky,70 and Mikolowsky and McMahon.71

Measurements on round or plane, heated or unheated, turbulent jets in which the main emphasis was on studying the turbulence structure in terms of Reynolds stresses, turbulent heat flux, or higher-order moments were carried out by

Presented as Paper 88-3725 at the 1st National Fluid Dynamics Congress, Cincinnati, OH, July 24-28, 1988; received Dec. 2, 1988; revision received and accepted for publication April 4, 1990. Copyright © 1990 by the American Institute of Aeronautics and Astronautics, Inc. All rights reserved.

^{*}Assistant Professor, Department of Mechanical Engineering, P.O. Box 248294. Member AIAA.

[†]Professor, Department of Mechanical Engineering. Member AIAA.

Crabb, ⁷² Crabb et al. ⁷³ Andreopoulos, ⁷⁴⁻⁷⁶ Andreopoulos and Rodi, ⁷⁷ Ramaprian and Haniu, ⁷⁸ Wu et al., ⁷⁹ and Sherif and Pletcher. ^{80,81}

Although the measurements reported in the literature seem to be numerous, few studies went beyond investigating the jet trajectory and spread. Measurements dealing with the mean temperature field are generally fewer than those on the velocity field. With the exception of the work by Andreopoulos, ⁷⁴ Ramaprian and Haniu⁷⁸ (two-dimensional jets), and Sherif and Pletcher, ⁸¹ there is no work known to the authors that reported measurements on the root-mean-square temperature fluctuations. The measurements reported by Andreopoulos, ⁷⁴ however, are applicable to low velocity ratio jets (less than 2.0).

Mechanism of Interaction Between a Jet and a Cross Section

According to measurements and flow visualization studies conducted by previous investigators, a fairly detailed description of the jet can be presented. When a jet is discharged vertically into a crossflowing stream, three distinct regions can be distinguished.

1) The initial region, or the zone of flow establishment (also called the zone of residual inlet velocity), 29 where the initially uniform jet flow interacts with the ambient crossflow causing a shear layer to develop at the jet boundaries. A cone-shaped cross section in this zone is formed where the turbulence intensity is relatively small, the velocity is nearly uniform, and the flow is nearly irrotational.³⁰ Because of a slight jet curvature in this zone, the cone-shaped potential core tends to deflect (see Fig. 1). At the upstream region of the jet, the flow is decelerated and a positive pressure region is formed. 16 Severe lateral shear stresses, directed toward the jet wake, act on the jet sides. A region of low pressure is formed in the jet wake, and separation there ultimately occurs. The length of the jet potential core is generally a function of the jet diameter and velocity ratio. Pratte and Baines¹⁹ argue that it is also a function of the jet discharge Reynolds number. Keffer and Baines¹⁷ also confirm this observation, but no functional trend is reported in the literature for this Reynolds number dependency.

2) The main region, or the zone of established flow (also called the zone of accommodation²⁹ and the curvilinear zone).³⁰ In this region a turbulent mixing layer develops around the jet boundaries and the flow quickly becomes fully turbulent.⁸² The jet sides experience strong lateral deflections because of the shearing action of the cross stream. The poten-

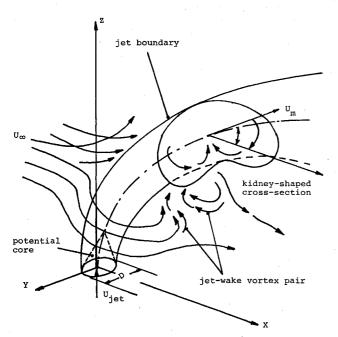


Fig. 1 Round turbulent jet in a crossflow.

tial core rapidly erodes primarily because of the growth of the mixing layer along the jet flow direction. Keffer⁸² reports that the length of the potential core is a direct function of the velocity ratio and is of the order of 1, 2, and 3 orifice diameters for velocity ratios of 4, 6, and 8, respectively.

Jordinson¹³ observed that the flow in this region is similar to that over a porous cylinder with suction. He reports that, in the jet wake, the main stream fluid is drawn into the cylinder so that the angles of the flow to the main stream direction are greater than 90 deg. The analogy given by Jordinson is probably not complete and does not adequately describe the entrainment mechanism at the jet front and in the wake. A pair of vortices is formed in the wake region which expands and intensifies by receiving vorticity produced by the interaction between the freestream and the jet.30 The jet cross-sectional shape is changed to a characteristic kidney shape by the action of the pressure and lateral shear forces. The jet sides are easily deflected since they possess less momentum than the jet center because of the lateral mixing process.¹⁷ The result is a pair of counter-rotating swirls, the strength of which reaches a maximum at some point downstream from the potential core region.³⁰ The helicity in the wake region contributes greatly to the mechanism by which the freestream fluid is entrained into the jet fluid.19

3) The far-field region (also known as the zone of velocity profile similarity)²⁹ where the magnitude and direction of the jet velocity are close to those of the crossflowing stream, and it becomes more and more difficult to distinguish freestream and jet fluids. The jet, however, will continue to rise, partly because the vertical component of the jet original momentum is preserved.82 The rate of rise, however, will continue to decrease primarily because the diffusion of the jet momentum takes place over a continuously increasing area. The two vortices will continue to move at nearly the velocity of the freestream, their strength will decrease, and their size will continue to increase. 30,58,59 The decrease in the amount of circulation is partly attributed to the increase in the rate of ordinary viscous dissipation. Keffer82 reports that the jet turbulence structure in this region is dominated by the viscous decay. He argues that the vortices in this phase should be viewed as general circulation patterns rather than discrete line vortices. The measurements of Pratte and Baines¹⁹ indicate that the entrainment starts to decrease and that the entrainment mechanism becomes entirely dependent on the vortex flow.

Results of more recent investigations^{73,75-77} at low velocity ratios indicate that the upstream flow in the jet dischage pipe is influenced by the crossflowing stream. For example, Crabb et al.⁷³ report that the distortion in the velocity profile at the pipe exit is more pronounced at a velocity ratio of 1.15 than at 2.3. For both ratios, however, the front half of the jet in the plane of symmetry has a decreased velocity, and the rear half is forced to accelerate, and probably widen, to compensate for the extra flow of jet fluid. Crabb et al. report that this rearedge acceleration is more noticeable in the lower velocity case.

This paper reports on mean and fluctuating temperature measurements in a heated nonbuoyant jet for velocity ratios of 1, 2, 4, and 7. The measurements are presented in the form of contours of mean and fluctuating temperatures both across and parallel to the jet. The jet and wake cross sections are revealed, showing regions of low and high temperature fluctuations and their relationship to the flow. Throughout the paper, reference is made to other work describing the jet-wake interaction in terms of other parameters such as pressure, velocity, turbulence, and temperature.

Experimental Program

As mentioned earlier, the measurements were carried out in the 0.61-m \times 1.067-m semiclosed circuit water channel at Iowa State University. The test section was formed by an aluminum plate, 12.7 mm thick, 0.457 m wide, and 1.829 m long with a rounded (1:2 ellipse) leading-edge and a trailing-edge flap. The plate was positioned 152.4 mm above the bottom of

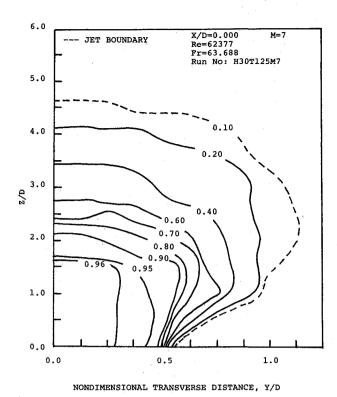


Fig. 2a Contour plots of $(T - T_{\infty})/(T_j - T_{\infty})$, M = 7, X/D = 0.

the channel. The plate length was chosen taking into account the range of jet trajectories for different jet inlet conditions. The plate width was chosen so as not to interfere with the turbulent boundary layers already developed on the channel sides. Two false sidewalls having a thickness of 8.73 mm, a height of 0.965 m, and the same plate length of 1.829 m were installed on both sides of the plate to promote two dimensionality for the flow and to serve as supporting members for the plate. Hot water was injected vertically upward from the plate through a circular pipe of 13.84 mm i.d. The thickness of the boundary layers developing on the false walls has been kept to a minimum by providing streamlined leading edges to the false walls. The flap provided at the plate trailing edge was adjusted so as to minimize the elliptic effects in the flowfield and to ensure that the flow approached the plate at a zero angle of incidence. A DISA constant current anemometer was used in conjunction with a DISA 55R11 fiber-film probe for mean and fluctuating temperature measurements. The constant-current anemometer was comprised of a DISA 55M01 main unit, a DISA 55M05 power pack, and a DISA 55M20 temperature bridge. A constant probe current of 2.5 mA was used for all runs. This relatively high value of the probe current was chosen because of the desire to increase probe sensitivity to temperature fluctuations. The velocity dependence was checked and was found to be negligible at this value of the probe sensor current. The DISA 55R11 probe was calibrated for temperature measurements using a Haake A81 constant temperature bath with a resolution of 0.056°C and a calibrated copper-constantan thermocouple attached to the fiberfilm probe support. Additional details about the calibration method, experimental facility, data acquisition system, and measurement techniques can be found in Sherif and Pletcher. 80,81 The overall uncertainties in the measurement of the mean temperature and root-mean-square temperature fluctuations were less than 2 and 12%, respectively.

Qualification of the measurement techniques was done in part by comparisons with mean temperature data of those of Ramsey and Goldstein, 50 Kamotani and Greber, 51 and Andreopoulos, 74-Other qualification tests can be found in Sherif and Pletcher, 80

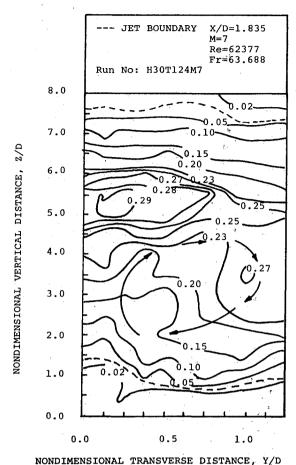


Fig. 2b Contour plots of $(T - T_{\infty})/(T_j - T_{\infty})$, M = 7, X/D = 1.835.

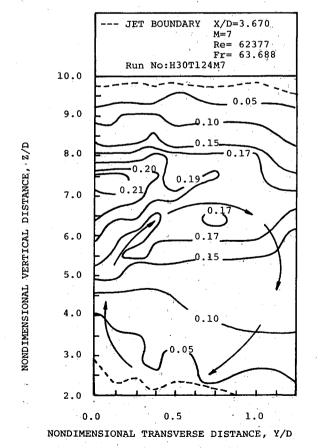


Fig. 2c Contour plots of $(T - T_{\infty})/(T_j - T_{\infty})$, M = 7, X/D = 3.670.

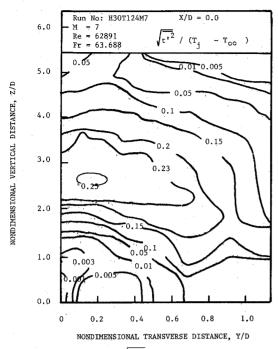


Fig. 3a Contour plots of $\sqrt{\overline{t'^2}}$ / $(T_j - T_\infty)$, M = 7, X/D = 0.

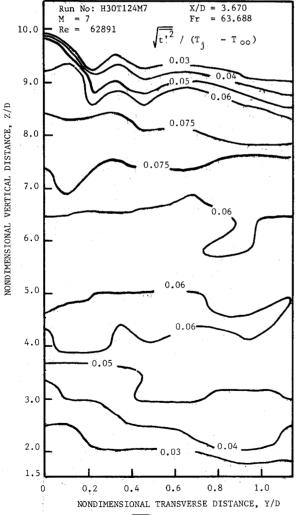
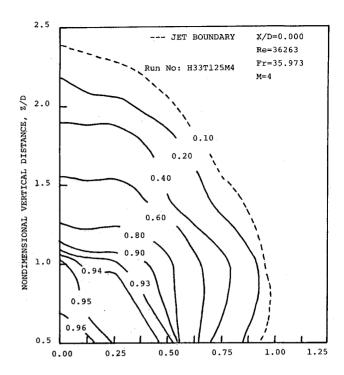


Fig. 3b Contour plots of $\sqrt{t^{\prime 2}}$ / $(T_j - T_{\infty}), M = 7, X/D = 3.67$.

Results and Discussion

As mentioned earlier, the jet/wake thermal characteristics will be presented in terms of contours of mean and fluctuating



NONDIMENSIONAL TRANSVERSE DISTANCE, Y/D

Fig. 4 Contour plots of $(T - T_{\infty})/(T_j - T_{\infty})$, M = 4, X/D = 0.

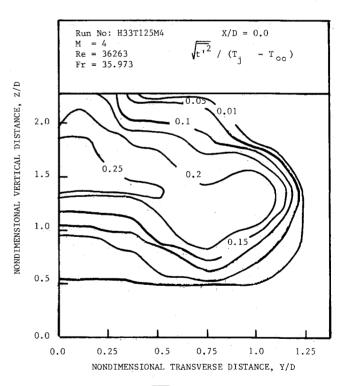


Fig. 5 Contour plots of $\sqrt{t^2}$ / $(T_j - T_\infty)$, M = 4, X/D = 0.

temperatures both across and parallel to the jet. In a previous paper, 81 the authors examined mean and fluctuating temperature profiles for velocity ratios of 1, 4, and 7. In that paper it was established that the M=1 jet behaved distinctly differently than the M=4 and M=7 jets. In order to examine this phenomenon more closely, the authors decided to study the M=2 jet. This paper reports in part on that.

For M=7, isotherms in cross planes (perpendicular to the jet plane of symmetry) are presented in Figs. 2a, 2b, and 2c for downstream positions of X/D=0, 1.835 and 3.670, respectively.

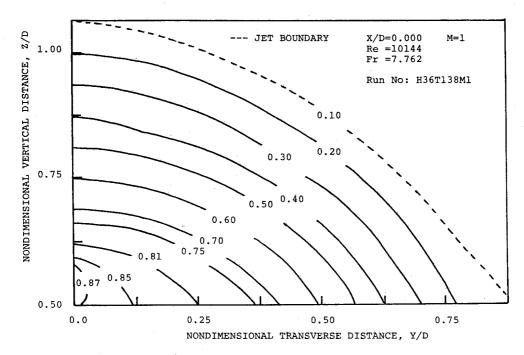


Fig. 6a Contour plots of $(T - T_{\infty})/(T_j - T_{\infty})$, M = 1, X/D = 0.

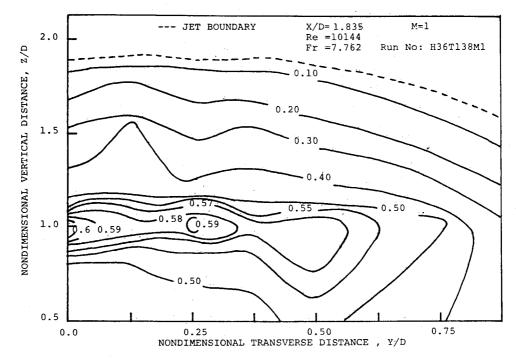


Fig. 6b Contour plots of $(T - T_{\infty})/(T_j - T_{\infty})$, M = 1, X/D = 1.835.

Above the jet discharge plane X/D=0 (see Fig. 2a) the isotherms appear as more or less concentric circles wih the highest temperature being close to the center. Slightly farther downstream, at X/D=1.835 (see Fig. 2b), the isotherms appear slightly distorted. The highest temperature is shown to correspond to a vertical position of Z/D=5.25. This was shown earlier⁸¹ to be the location of the jet temperature centerline trajectory. Below Z/D=4.3, the isotherms look more nonuniform, indicating a region of more activity (mixing). According to Keffer, ⁸² the zone of flow establishment should end around X/D=2.5 for this velocity ratio. The isotherms displayed in Fig. 2b are therefore representative of the jet/wake thermal behavior in this zone.

At a downstream position of X/D = 3.67 (see Fig. 2c), more nonuniformities in the isotherms are noticeable. The jet

centerline is approximately located at a vertical distance of Z/D=7.4 with a local temperature excess ratio about 21% of that at the discharge. This downstream position is about 1 diam into the second zone (zone of established flow) according to Keffer.⁸²

In all three figures, dashed lines indicating the location of the jet upper and lower edges are shown. The jet edge as presented here is based on a concept similar to that introduced by Squire. §3 According to Squire, the jet edge is defined as the boundary on which the velocity is 10% of the centerline value. In the present investigation, a similar definition was adopted employing the temperature excess ratio in place of velocity. The jet edge as introduced here, however, should be taken to imply both the jet and its wake. This implies that within the "jet boundaries" most of the activities will take place and that

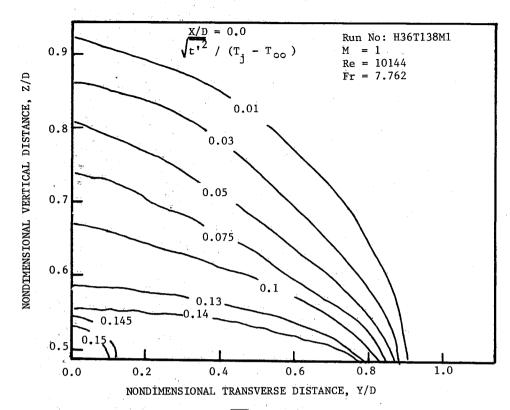


Fig. 7a Contour plots of $\sqrt{t^2}$ / $(T_j - T_\infty)$, M = 1, Y/D = 0.

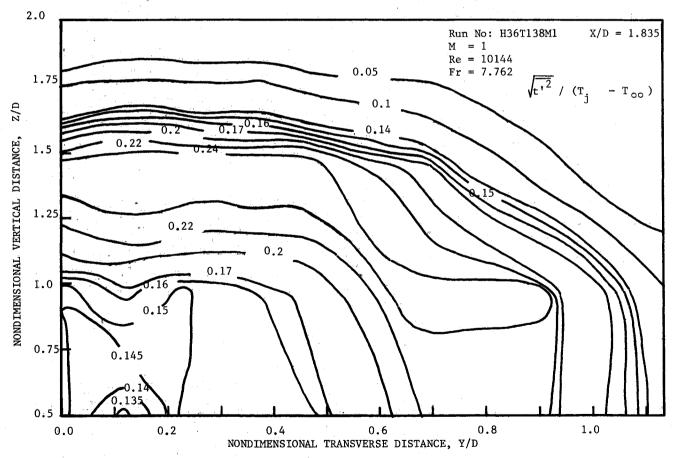


Fig. 7b Contour plots of $\sqrt{\overline{t'^2}}$ / $(T_j - T_\infty)$, M = 1, X/D = 1.835.

the regions not included within the boundaries are mostly freestream or low-activity regions. It will be shown a little later, however, that that may not be quite true for a low velocity ratio jet: The temperature fluctuation contours for the same velocity ratio and two representative downstream positions are presented in Figs. 3a and 3b, respectively. Additional contours of temperature fluctuations can be found in Sherif.⁸⁴

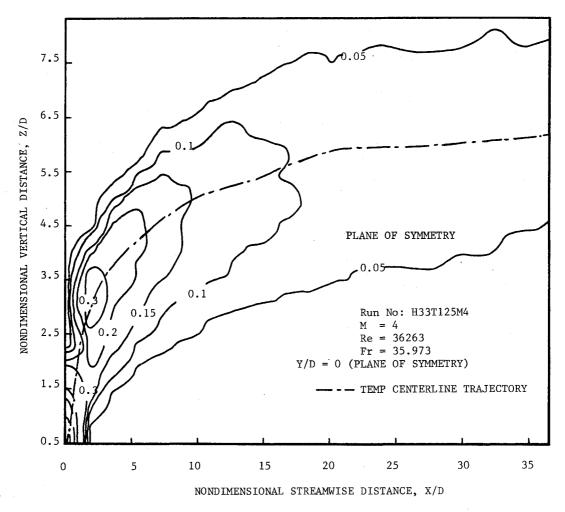


Fig. 8 Contour plots of $(T - T_{\infty})/(T_j - T_{\infty})$, M = 4, Y/D = 0.

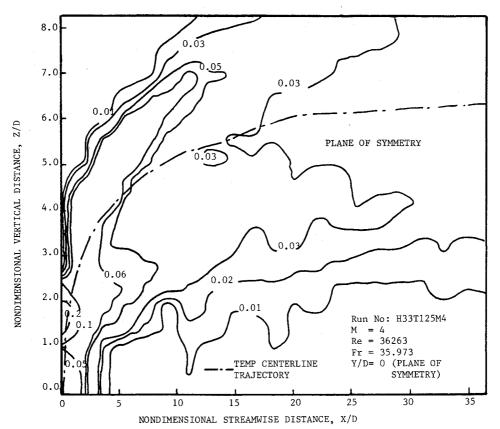


Fig. 9 Contour plots of $\sqrt{\overline{t'^2}}$ / $(T_j - T_\infty)$, M = 4, Y/D = 0.

Examination of Figs. 3a and 3b indicates that the fluctuating temperature exhibits a rather different behavior than the mean temperature. Above the jet discharge plane (X/D = 0), fluctuations reach a maximum at about a vertical position of 2.6. Referring to Fig. 2a, it becomes apparent that the maximum fluctuations occur above the jet centerline in a region of high mean temperature gradient. Referring to Fig. 2c, it is easily seen that the mean temperature excess ratio drops from 0.9 to 0.6 when the vertical distance is increased from 2.2 to 2.75, respectively. This results in a temperature gradient of 0.546. Similarly, between Z/D = 2.75 and Z/D = 4.6, the temperature excess ratio drops from 0.6 to 0.1, resulting in a gradient of 0.27, which is about half the temperature gradient immediately above the jet discharge. The results are also consistent with the findings of Keffer,82 who observed that the turbulence intensity in the potential core is small. For example, for vertical distances Z/D smaller than 0.5, a temperature fluctuation of only 0.1% of the temperature excess at the discharge is observed (see Fig. 3a). This compares with a maximum value of 25% at about Z/D of 2.75. This implies that the temperature fluctuation behaves rigorously in the same manner as the turbulence intensity. It may not be inaccurate therefore to refer to the former as a turbulent temperature.

The authors⁸¹ have established that the fluctuating temperature (like many other turbulent quantities)⁸⁰ exhibits a double-peak pattern. The lower peak is smaller and usually occurs in the wake, whereas the higher peak is larger and occurs at a region of maximum temperature gradient above the jet centerline. This behavior is rather difficult to see by merely examining the contour plots and is best illustrated by examining the fluctuating temperature profiles.

At a velocity ratio of M=4, a more or less similar behavior is exhibited by the jet. Mean and fluctuating temperature contours for X/D=0 are presented in Figs. 4 and 5, respectively. Additional contours at M=4 can be found in Sherif and Pletcher.⁸¹

At a velocity ratio of 1, the jet appears to exhibit a different behavior. This can be demonstrated by examining contours of mean and fluctuating temperature at X/D=0 and 1.835 in Figs. 6a and 6b, and 7a and 7b, respectively. Contours at positions farther downstream can be found in Sherif.⁸⁴

Immediately above the jet discharge plane (X/D=0), both mean and fluctuating temperature contours are remarkably uniform (Figs. 6a and 7a). Farther downstream at X/D=1.835, the jet centerline rises as indicated by relatively large temperature excess ratios at higher vertical positions. For ex-

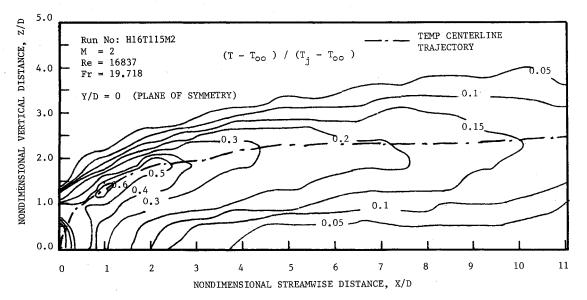


Fig. 10 Contour plots of $(T - T_{\infty})/(T_j - T_{\infty})$, M = 2, Y/D = 0 (plane of symmetry).

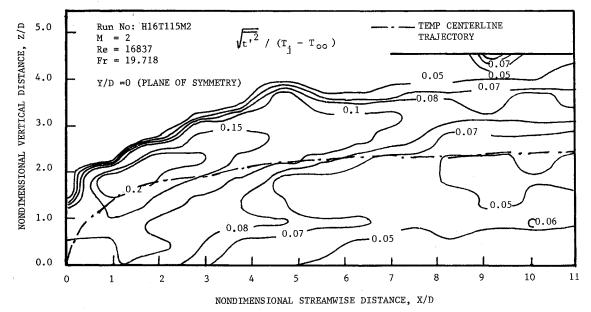


Fig. 11 Contour plots of $\sqrt{t^2}$ / $(T_i - T_\infty)$, M = 2, Y/D = (plane of symmetry).

ample, Fig. 6b indicates a maximum temperature excess ratio in the jet plane of symmetry (Y/D = 0) of 0.6 at about a Z/Dof 1. The contours in Fig. 6b in the jet core look more or less like a kidney, confirming the existence of the well-known kidney-shaped region. It is interesting to note, however, that the wake region is very small (almost nonexistent) as evidenced by high values of the temperature excess ratio near the jet discharge plane (Z/D = 0). The same trend continues at positions farther downstream (see Sherif⁸⁴). Above the jet centerline, the contours exhibit a uniform appearance indicating a region of freestream fluid and minimum mixing.

The fact that the wake region is small for this velocity ratio suggests that freestream fluid is not sucked underneath the jet as it would be in a larger velocity ratio jet. As a consequence, it is believed that the entrainment mechanism is not as efficient in the former case and that the fluid underneath the jet is mostly jet fluid (not freestream). These observations are reaffirmed by examining the temperature fluctuation data presented in Fig. 7b for the same jet.

The small velocity ratio jet is further different in terms of the double-peak pattern for the turbulent quantities.80,81 For the fluctuating temperature, one can observe only one peak that occurs in a region of high temperature gradient above the centerline temperature trajectory.

The far-field zone is best illustrated through examination of contour plots in the jet plane of symmetry. This is done for velocity ratios of 4 and 2 in Figs. 8 or 9 for M = 4 and Figs. 10 and 11 for M = 2.

Figures 8 and 10 for the mean temperature contours reveal an interesting fact about the jet in crossflow. It is obvious that this type of flow is asymmetric with the jet temperature axis being curved toward the lower boundary of the jet. Previous investigators have observed this phenomenon and attributed it to the presence of a pressure field associated with the obstacle formed by the jet near the discharge. This pressure field is characterized by a low pressure region upstream of the jet front boundaries (see Chassaing et al.²⁹).

The fluctuating temperature contours in the plane of symmetry, on the other hand, reveal the regions of high and low fluctuations up to X/D = 37. It is easy to see that the region around the centerline trajectory is indeed a region of low fluctuations. For example, for M = 4, the region of maximum fluctuations is mostly bounded by a \mathbb{Z}/\mathbb{D} of 1.2 and 2 and X/D of 0 and 1.5. In this region, fluctuations on the order of 20-23% of the temperature excess at discharge are observed. About 15 diam downstream close to the jet centerline, fluctuations on the order of 3% are seen. Still, the effect of the jet can be felt as far as 37 jet diam downstream. Ades⁸⁵ observed a jet as far as 100 diam downstream.

The fluctuation contours are most uniform around the jet centerline. Away from the centerline and particularly in the wake region, the contours are most nonuniform. The magnitude continues to decrease with increasing downstream distance primarily because of heat transferred to the freestream fluid.

The contour plots presented in Figs. 10 and 11 for M = 2reveal some interesting observations. Like the M = 1 jet, the wake region is small as relatively high temperatures are observed near the solid boundary. It is not inaccurate to assume that the M = 2 jet is a borderline case between the low and high velocity ratio jets. In other words, one should expect the M = 2 jet to exhibit characteristics similar to both the M = 4 and M = 1 jets. The validity of this assertion can be checked by examining Figs. 10 and 11 and comparing them with the corresponding figures for M = 4 and M = 1.

Conclusions

A fairly detailed picture of the jet-wake thermal characteristics is presented for different velocity ratios and downstream positions. The behavior of the jet was described by examining contours of mean and fluctuating temperatures both across and parallel to the jet plane of symmetry. It was established

that a small velocity ratio jet behaves differently than a large velocity ratio jet (M > 2). The existence of a double vortex structure was confirmed in both jets but was found to be weaker in a small velocity ratio jet. Thus, it was concluded that the entrainment and mixing mechanisms are less effective in the latter case. The M = 2 jet is believed to be a borderline case between the two jets.

Acknowledgments

This material is based on work supported by the National Science Foundation under Grants ENG-7812901 and MEA-8211713. Support from the Graduate College and the Department of Mechanical Engineering at Iowa State University is also gratefully acknowledged.

References

¹Rodi, W., Turbulent Buoyant Jets and Plumes, Pergamon, New York, 1982.

²Bryant, L. W., and Cowdrey, C. F., "Effects of Velocity and Temperature of Discharge on the Shape of Smoke Plumes from a Funnel or Chimney: Experiments in a Wind Tunnel," *Proceedings of* the Institution of Mechanical Engineers, Suffolk, UK, Vol. 169, 1955,

pp. 371-384.

³Slawson, P. R., and Csanady, G. T., "On the Mean Path of Buoyant, Bent-Over Chimney Plumes," Journal of Fluid Mechanics, Vol. 28, Pt. 2, 1967, pp. 311-322.

⁴Halitsky, J., "Single Camera Measurements of Smoke Plumes,"

International Journal of Air and Water Pollution, Vol. 4, 1961, pp.

⁵Hewett, T. A., Fay, J. A., and Hoult, D. P., "Laboratory Experiments of Smokestack Plumes in a Stable Atmosphere," Atmospheric Environment, Vol. 5, Pergamon, New York, 1971, pp. 767-789.

⁶Hewett, T. A., "Model Experiments of Smoke-Stack Plumes in a Stable Atmosphere," Ph.D. Dissertation, Massachusetts Inst. of Technology, Cambridge, MA, 1970.

⁷Bringfelt, B., "Plume Rise Measurements at Industrial Chimneys," Atmospheric Environment, Vol. 2, Pergamon, New York, 1968, pp. 575-598.

⁸Leavitt, J. M., Carpenter, S. B., and Thomas, F. W., "An Interim Report on Full-Scale Study of Plume Rise at Large Generating Stations," APCA Annual Meeting, Air Pollution Control Association, Pittsburgh, PA, June 1985.

⁹Carpenter, S. B., Thomas, F. W., and Gastrell, F. E., "Full-Scale Study of Plume Rise at Large Electric Generating Stations," Tennessee Valley Authority, Knoxville, TN, Sept. 1968.

10 Thomas, F. W., Carpenter, S. B., and Colbaugh, W. C., "Plume Rise Estimates for Electric Generating Stations," *Philosophical* Transactions of the Royal Society of London, Vol. A-265, No. 1161, 1969, pp. 221-243.

¹¹Weil, J. C., "Model Experiments of High Stack Plumes," M.S. Thesis, Massachusetts Inst. of Technology, Cambridge, MA 1968.

¹²Barilla, P. A., "Dependence of Entrainment Coefficient upon Orifice Conditions in Model Studies of a Smoke Plume in a Laminar Cross Wind," M.S. Thesis, Massachusetts Inst. of Technology, Cambridge, MA, 1968.

¹³Jordinson, R., "Flow in a Jet Directed Normal to the Wind," British Aeronautical Research Council, R & M 3074, 1956.

¹⁴Shandorov, G. S., "Flow from a Channel into Stationary and Moving Media," Zhurnal Tekhnicheskoi Fiziki, Vol. 37, No. 1, 1957,

pp. 1.

15 Gordier, R. L., "Studies on Fluid Jets Discharging Normally into Moving Liquid," St. Anthony Falls Hydraulics Lab., Univ. of Minne-

sota, Minneapolis, MN, Paper 28-B, 1969.

16Keffer, J. F., "The Round Turbulent Jet in a Cross-Wind," Ph.D. Dissertation, Univ. of Toronto, Toronto, Canada, 1962.

17Keffer, J. F., and Baines, W. D., "The Round Turbulent Jet in a

Cross Wind," Journal of Fluid Mechanics, Vol. 15, Pt. 4, 1963, pp. 481-497.

¹⁸Pratte, B. D., "Profiles of the Round Turbulent Jet in a Cross Wind," M.A. Sc. Thesis, Dept. of Mechanical Engineering, Univ. of Toronto, Toronto, Canada, 1964.

¹⁹Pratte, B. D., and Baines, W. D., "Profiles of the Round Turbulent Jet in a Cross Flow," Journal of the Hydraulics Division, Proceedings of the ASCE, HY6, Vol. 93, 1967, pp. 53-64; corrections, HY3, Vol. 94, May 1968, pp. 815, 816.

²⁰Gerend, R. P., "Penetration of a Jet into a Non-Uniform Stream," M.S. Thesis, Seattle Univ., Seattle, WA, 1968.

21 Patrick, M. A., "Experimental Investigation of the Mixing and

Penetration of a Round Turbulent Jet Injected Perpendicularly into a Traverse Stream," Transactions of the Institution of Chemical Engineers, Vol. 45, No. 1, 1967, pp. 16-31.

²²Platten, J. L., and Keffer, J. F., "Deflected Turbulent Jet Flows," ASME Journal of Applied Mechanics, Vol. 38, Dec. 1971,

pp. 756-758.

²³Margason, R. J., "The Path of a Jet Directed at Large Angles to a Subsonic Free Stream," NASA Technical Notes, No. D-9419, Nov.

²⁴Bergeles, G., Gosman, A. D., and Launder, B. E., "The Near-Field Character of a Jet Discharged through a Wall at 90 Deg. to a Main Stream," American Society of Mechanical Engineers, New York, ASME Paper 75-WA/HT-108, 1975.

²⁵Storms, K. R., "Low-Speed Wind Tunnel Investigation of a Jet Directed Normal to the Wind," Aeronautics Lab., Univ. of Washing-

ton, Seattle, WA, Rept. No. 995, Nov. 1965.

26 Gelb, G., and Martin, W., "An Experimental Investigation of the Flow Field about a Subsonic Jet Exhausting into a Low Velocity Airstream," Northrop Norair, Los Angeles, Rept. NOR65-229, Aug.

²⁷Gelb, G., and Martin, W., "An Experimental Investigation of the Flow Field about a Subsonic Jet Exhausting into a Quiescent and a Low Velocity Air Stream," Canadian Aeronautics and Space Journal, Vol. 12, No. 8, 1966, pp. 333-342.

²⁸Crowe, C. T., and Riesebieter, H., "An Analytical and Experimental Study of Jet Deflection in a Cross Flow," Advisory Group for Aerospace R & D Preprint, Paris, 1967.

²⁹Chassaing, P., George, J., Claria, A., and Sananes, F., "Physical Characteristics of Subsonic Jets in a Cross-Stream," Journal of Fluid

Mechanics, Vol. 62, Pt. 1, 1974, pp. 41-64.

³⁰Chan, T., and Kennedy, J. F., "Turbulent Non-Buoyant or Buoyant Jets Discharged into Flowing or Quiescent Fluids," Iowa Institute of Hydraulics Research, Rept. No. IIHR 140, Univ. of Iowa, Iowa City, IA, Aug. 1972.

³¹Fan, L. N., "Turbulent Buoyant Jets into Stratified or Flowing Ambient Fluids," Ph.D. Dissertation, Lab. for Hydraulic and Water

Resources, California Inst. of Technology, Pasadena, CA, 1967.

32Fan, L. N., "Turbulent Buyoant Jets into Stratified or Flowing Ambient Fluids," Lab. for Hydraulic and Water Resources, Califor-

nia Inst. of Technology, Pasadena, CA, Rept. KH-R-15, 1967.

33Abraham, G., and Eysink, W. D., "Jets Issuing into Fluid with a Density Gradient," Journal of Hydraulic Research, Vol. 7, No. 2, 1969, pp. 145-175.

³⁴Abraham, G., "The Flow of Round Buoyant Jets Issuing Vertically into Ambient Fluid Flowing in a Horizontal Direction," Advances in Water Pollution Research, edited by S. H. Jenkins, Pergamon, New York, 1971.

35 Turner, J. S., "Jets and Plumes with Negative or Reversing Buoy-

ancy," Journal of Fluid Mechanics, Vol. 26, 1966, pp. 779-792.

Vadot, M. L., "Etude de la Diffusion des Panaches de Fumee dans L'Atmosphere," Centre International Technique d'Etudes Pollution Atmospherique, Paris, 1965.

³⁷Ayoub, G. M., "Dispersion of Buoyant Jets in a Flowing Ambient Fluid," Ph.D. Dissertation, Univ. of London, Dept. of Civil Engineering, March 1971.

Cavola, R. G., "An Experimental/Analytical Investigation of Negatively Buoyant Jets Discharged Vertically Upward into a Crossflow Current," M.S. Thesis, Dept. of Mechanical Engineering, Ore-

gon State Univ., Corvallis, OR, 1982.

39 Cavola, R. G., and Davis, L. R., "An Experimental Investigation of Negatively Buoyant Jets Discharged into a Cross Flow," American Society of Mechanical Engineers, New York, ASME Paper 83-WA-HT-105, 1983.

⁴⁰Callaghan, E. E., and Ruggeri, R. S., "Investigation of the Penetration of an Air Jet Directed Perpendicularly to an Air Stream,'

NACA Technical Notes, No. 1615, June 1948.

41 Callaghan, E. E., and Ruggeri, R. S., "A General Correlation of Temperature Profiles Downstream of a Heated Air Jet Directed Perpendicularly to an Air Stream," NACA Technical Notes, No. 2466,

Sept. 1951.

42Callaghan, E. E., and Bowden, D. T., "Investigation of Flow Coefficient of Circular, Square, and Elliptical Orifices at High Pressure Ratios," *NACA Technical Notes*, No. 1947, 1949.

43 Ruggeri, R. S., and Callaghan, E. E., "Penetration of Air Jets Is-

suing from Circular, Square, and Elliptical Orifices Directed Perpendicularly to an Air Stream," NACA Technical Notes, No. 2019, Feb.

1950.

44Ruggeri, R. S., "General Correlation of Temperature Profiles

15 Directed at Various Angles to an Air Stream," NACA Technical Notes, No. 2855, Dec. 1952.

⁴⁵Campbell, J. F., and Schetz, J. A., "Penetration and Mixing of Heated Jets in a Waterway with Application to the Thermal Pollution Problem," AIAA Paper 71-524, May 1971.

⁴⁶Campbell, J. F., and Schetz, J. A., "Analysis of the Injection of a Heated Turbulent Jet into a Moving Mainstream, with Emphasis on a Thermal Discharge in a Waterway," Virginia Polytechnic Inst. and State Univ., Blacksburg, VA, Rept. No. VPI-E-72-24, Dec. 1977.

⁴⁷Campbell, J. F., and Schetz, J. A., "Flow Properties of Submerged Heated Effluents in a Waterway," AIAA Paper 72-79, Jan. 1972; also *AIAA Journal*, Vol. 11, No. 2, 1973, pp. 223-230. ⁴⁸Payne, J. A., and Schetz, J. A., "An Experimental Comparison

of Thermal Discharges to a Waterway by a Single Jet and by a Multiple Port Diffuser," Virginia Polytechnic Inst. and State Univ.,

Blacksburg, VA, Rept. No. VPI-E-73-21, Sept. 1974.

49Ramsey, J. W., "The Interaction of a Heated Jet with a Deflecting Flow," Ph.D. Dissertation, Univ. of Minnesota, Minneapolis,

MN, 1969. 50Ramsey, J. W., and Goldstein, R. J., "Interaction of a Heated Jet with a Deflecting Stream," NASA CR-92613, April 1970.

⁵¹Kamotani, Y., and Greber, I., "Experiments on a Turbulent Jet in a Cross Flow," NASA CR-72893, 1971; also AIAA Journal, Vol.

 No. 11, 1972, pp. 1425-1429.
 Wark, C. E., and Foss, J. F., "Thermal Measurements for Jets in Disturbed and Undisturbed Crosswind Conditions," AIAA Journal,

Vol. 26, No. 8, 1988, pp. 901, 902.

⁵³Chang-Lu, and Hsiu-Chen, "Aufrollung Eines Zylindrishen Strahles Durch Querwind," Doctoral Dissertation, Univ. of Gottingen, Germany, 1942.

54Rouse, H., "Diffusion in the Lee of a Two-Dimensional Jet,"

Proceedings of the 9th Congress of Applied Mechanics, Vol. 1, Univ. of Brussels, Brussels, Belgium, 1957, pp. 307-315.

55McAllister, J. D., "A Momentum Theory for the Effects of Cross

Flow on Incompressible Turbulent Jets," Ph.D. Dissertation, Univ.

of Tennessee, Tullahoma, TN, Aug. 1968.

56Pratte, B. D., and Keffer, J. F., "Swirling Turbulent Jet Flows,
Part I: The Single Swirling Jet," Dept. of Mechanical Engineering, Univ. of Toronto, Toronto, Canada, Rept. No. TP-6901, March 1969.

⁵⁷McMahon, H. M., Hester, D. D., and Palfrey, J. G., "Vortex Shedding from a Turbulent Jet in a Cross-Wind," Journal of Fluid

Mechanics, Vol. 48, Pt. 1, 1971, pp. 73-80.

58 Fearn, R., and Weston, R. P., "Vorticity Associated with a Jet in a Cross Flow," AIAA Journal, Vol. 12, No. 12, 1974, pp. 1666-1671.

59 Weston, R. P., "A Description of the Vortex Pair Associated with a Jet in a Cross-Flow," Workshop Proceedings: Prediction Methods for Jet/VSTOL Propulsion Aerodynamics, Institute for Defense Analysis, Alexandria, VA, 1975.

⁶⁰Moussa, Z. M., "The Near Field Flow of an Axisymmetric Jet in a Cross-Stream," Ph.D. Dissertation, Syracuse Univ., Syracuse, NY,

1976.

⁶¹Moussa, Z. M., Trischka, J. W., and Eskinazi, S., "The Near Field in the Mixing of a Round Jet with a Cross-Stream," Journal of

Fluid Mechanics, Vol. 80, 1977, pp. 49-80.

62 Vogler, R. D., "Surface Pressure Distributions Induced on a Flat Plate by a Cold Air Jet Issuing Perpendicularly from the Plate and Normal to a Low-Speed Free-Stream Flow," NASA Technical Notes,

No. D1629, March 1963.

63 Peake, D. J., "The Pressures on a Surface Surrounding a Jet Issuing Normal to a Main Stream," National Research Council of Canada, Aerospace LR-410, Nov. 1964.

⁶⁴Bradbury, L. J. S., and Wood, M. N., "The Static Pressure Distribution Around a Circular Jet Exhausting Normally from a Plane Wall into an Airstream," British Aeronautical Research Council, London, C.P. No. 822, 1965.

65 Wooler, P. T., Burghard, G. H., and Gallagher, J. T., "Pressure Distribution on a Rectangular Wing with a Jet Exhausting Normally into an Air Stream," Journal of Aircraft, Vol. 4, No. 6, 1967, pp.

537-543.

66Hackett, J. E., and Miller, H. R., "The Aerodynamics of the Lifting Jet in a Cross-Flowing Stream," Analysis of a Jet in a Subsonic Crosswind, NASA SP-218, 1969, pp. 37-48.

⁶⁷McMahon, H. M., and Mosher, D. K., "Experimental Investigation of Pressures Induced on a Flat Plate by a Jet Issuing into a Subsonic Crosswind," Analysis of a Jet in a Subsonic Crosswind, NASA SP-218, 1969, pp. 49-62.

68 Mosher, D. K., "An Experimental Investigation of a Turbulent Jet in a Crossflow," Ph.D. Dissertation, Georgia Inst. of Technology, Atlanta, GA, 1970.

⁶⁹Wu, J. C., McMahon, H. M., Mosher, D. K., and Wright, M. A., "Experimental and Analytical Investigations of Jets Exhausting into a Deflecting Stream," Journal of Aircraft, Vol. 7, No. 1, 1970,

pp. 44-51.

70 Mikolowsky, W. T., "An Experimental Investigation of a Jet Issuing from a Wing in Crossflow," Ph.D. Dissertation, Georgia Inst. of Technology, Atlanta, GA, 1972.

⁷¹Mikolowsky, W. T., and McMahon, H., "An Experimental Investigation of a Jet Issuing from a Wing in a Crossflow," Journal of Aircraft, Vol. 10, No. 9, 1973, p. 546.

⁷²Crabb, D., "Jets in Cross Flow," Ph.D. Dissertation, Univ. of

London, London, 1979.

⁷³Crabb, D., Durao, D. F. G., and Whitelaw, J. H., "A Round Jet Normal to a Cross Flow," American Society of Mechanical Engineers Paper 80-WA/FE-10, 1980.

⁴Andreopoulos, J., "Heat Transfer Measurements in a Heated Jet-Pipe Flow Issuing into a Cold Cross-Stream," Physics of Fluids,

Vol. 26, No. 11, 1983, pp. 3201-3210.

75 Andreopoulos, J., "Measurements in a Pipe Flow Issuing Perpendicularly into a Cross Stream," Journal of Fluids Engineering, Vol. 104, No. 4, 1982, pp. 493-499.

76 Andreopoulos, J., "On the Structure of Jets in a Crossflow,"

Journal of Fluid Mechanics, Vol. 157, 1985, p. 163.

77 Andreopoulos, J., and Rodi, W., "Experimental Investigation of

Jets in a Crossflow," Journal of Fluid Mechanics, Vol. 138, 1984, pp. 93-127.

78 Ramaprian, B. R., and Haniu, J., "Turbulence Measurements in

Plane Jets and Plumes in Cross-flow," Iowa Inst. of Hydraulics Research, Univ. of Iowa, Iowa City, IA, Rept. No. IIHR 266, Aug.

⁷⁹Wu, J. M., Vakili, A. D., and Wu, F. M., "Investigation of the Interacting Flow of Nonsymmetric Jets in Crossflow," AIAA

Journal, Vol. 26, No. 8, 1988, pp. 940-947.

80 Sherif, S. A., and Pletcher, R. H., "Measurements of the Flow and Turbulence Characteristics of Round Jets in Cross Flow," AIAA Paper 86-1110, May 1986; also Journal of Fluids Engineering, Vol. 111, No. 2, 1989, pp. 165-171.

81 Sherif, S. A., and Pletcher, R. H., "Measurements of the Thermal Characteristics of Heated Turbulent Jets in Cross Flow," American Society of Mechanical Engineers Paper 86-HT-34, 1986; also Journal of Heat Transfer, Vol. 111, No. 4, 1989, pp. 897-903.

²Keffer, J. F., "The Physical Nature of the Subsonic Jet in a Cross-Stream," Analysis of a Jet in a Subsonic Crosswind," NASA SP-218, 1969, pp. 19-36.

83 Squire, H. B., "Jet Flow and its Effect on Aircrafts," Aircraft Engineering, Vol. 22, No. 1, 1950, p. 62.

Sherif, S. A., "Measurements of the Flow and Thermal Characteristics of Turbulent Jets in Cross Flow," Ph.D. Dissertation, Iowa State Univ., Ames, IA, 1985.

85 Ades, M., "Analysis of Round, Turbulent Buoyant Jets Discharged to a Cross Flow," M.S. Thesis, Israel Inst. of Technology, Israel, March 1975.

Recommended Reading from the AIAA Progress in Astronautics and Aeronautics Series . . . **GALA**



Dynamics of Flames and Reactive Systems and Dynamics of Shock Waves, **Explosions, and Detonations**

J. R. Bowen, N. Manson, A. K. Oppenheim, and R. I. Soloukhin, editors

The dynamics of explosions is concerned principally with the interrelationship between the rate processes of energy deposition in a compressible medium and its concurrent nonsteady flow as it occurs typically in explosion phenomena. Dynamics of reactive systems is a broader term referring to the processes of coupling between the dynamics of fluid flow and molecular transformations in reactive media occurring in any combustion system. Dynamics of Flames and Reactive Systems covers premixed flames, diffusion flames, turbulent combustion, constant volume combustion, spray combustion nonequilibrium flows, and combustion diagnostics. Dynamics of Shock Waves, Explosions and Detonations covers detonations in gaseous mixtures, detonations in two-phase systems, condensed explosives, explosions and interactions.

Dynamics of Flames and **Reactive Systems** 1985 766 pp. illus., Hardback ISBN 0-915928-92-2 AIAA Members \$59.95 Nonmembers \$92.95 **Order Number V-95**

Dynamics of Shock Waves, Explosions and Detonations 1985 595 pp., illus. Hardback ISBN 0-915928-91-4 AIAA Members \$54.95 Nonmembers \$86.95 Order Number V-94

TO ORDER: Write, Phone or FAX: American Institute of Aeronautics and Astronautics, c/o TASCO, 9 Jay Gould Ct., P.O. Box 753, Waldorf, MD 20604 Phone (301) 645-5643, Dept. 415 FAX (301) 843-0159

Sales Tax: CA residents, 7%; DC; 6%. Add \$4.75 for shipping and handling of 1 to 4 books (Call for rates on higher quantities). Orders under \$50.00 must be prepaid. Foreign orders must be prepaid. Please allow 4 weeks for delivery. Prices are subject to change without notice. Returns will be accepted within 15 days.